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THESIS

AIRCREW CENTERED SYSTEM DESIGN ANALYSIS CONSIDERATIONS FOR THE MH-53E HELICOPTER

by

Gregory J. Gibson

December, 1996

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AIRCREW CENTERED SYSTEM DESIGN ANALYSIS CONSIDERATIONS FOR THE MH-53E HELICOPTER

Gregory J. Gibson Lieutenant, United States Navy B.S., University of Missouri, 1988

Submitted in partial fulfillment of the requirements for the degree of

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ABSTRACT

An analysis was made of the aircrew centered system design aspects for the MH-53E helicopter. These aircrew centered design features included changes in the cockpit, aircraft weight and drag coefficient. The cockpit evaluation compared the current MH-53E cockpit configuration with design changes currently under review by the Navy. This evaluation suggests that the proposed cockpit design display change may reduce aircrew load stress and improve mission effectiveness. Changes in subsystem components may either increase or decrease the weight of the MH-53E. Similarly, changes in crew tasking may result in a need for more or less fuselage volume size. Therefore, the sensitivity of MH-53E performance to generic changes in weight and drag was investigated in order to make source assessment of equipment and crew tasking changes upon MH-53E mission effectiveness.

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I. INTRODUCTION

Through the advent of modern technology, aircraft crew stations provide aircrews with a variety of information, including information related to laser guided weapons, radar, etc. An effective crewstation requires displays, controls and avionics subsystems that enhance the mission effectiveness of the crew.

Today's military helicopters are often tasked to perform a multitude of missions. In addition to the variety of tasks, the complexity of that tasking can make for a challenging aircrew system design. To ensure that the crew is best able to perform these various and complex tasks, it is important that the aircrew system be designed to enhance the aircraft's performance as a weapon system. Thus, helicopter aircrew centered system design is essential.

Of all the missions performed by helicopter aircrews few are as uniquely complex as the Airborne Mine Countermeasures (AMCM). The focus of this thesis is the analysis of changes and suggested improvements in AMCM crewstations that may lead to an improvement in the effectiveness of the AMCM mission from a crew centered system design perspective.

II. THE MCM MISSION

A. GENERAL

There are primarily two ways to detect a sea mine, and one of them puts a hole in your ship. Mine Countermeasures (MCM) is the search, detection and neutralization of sea mines. The platforms by which the Navy accomplishes this mission is primarily performed by surface ships and helicopters. The ships that perform the MCM mission do it by primarily towing devices behind the ship that search or help neutralize mines (sonar, magnetic coils, etc). The MCM performed by helicopters is properly referred to as the Airborne Mine Countermeasures (AMCM) and is performed by the helicopter towing devices through the water which search or help neutralize mines (sonar, magnetic coils, etc). Since there are a variety of AMCM devices, for purposes of clarity they are collectively referred to as a "towed body." The AMCM towing subsystem uses a cable, with one end is attached internally to the aft end of the helicopter and the other end is attached to the "towed body."

The Airborne Mine Countermeasures (AMCM) is currently performed by the MH-53E helicopter. The AMCM mission is particularly unique in that it requires a substantial degree of crew coordination to facilitate mission success. Thus, the effective use of airframe systems to search detect and neutralize sea mine threats requires a thoroughly integrated aircrew centered system planform design.

B. DESCRIPTION OF THE AMCM MISSION

The following discussion provides a description of relevant AMCM parameters and mission aspects. The primary mission phases are composed of three consecutive phases of activity termed *streaming*, towing & recovery.

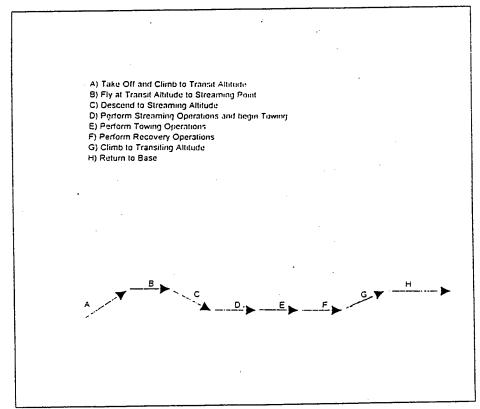


Figure 1 AMCM Mission Outline

For clarification Fig (1) depicts an outline of a typical AMCM mission which includes the previously mentioned mission phases.

1. Minesweeping and Minehunting

The AMCM mission is primarily composed of either one of two tasks: minesweeping or minehunting.

Minesweeping is the act of neutralizing mines or mine threats. Minesweeping is performed mechanically by towing cables equipped to cut the mooring chains of moored mines Fig (2).

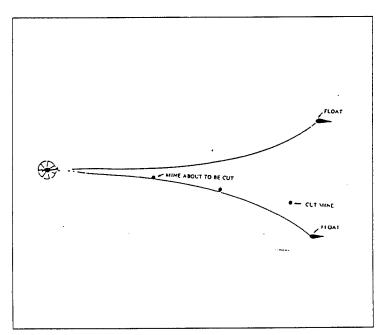


Figure 2 Mechanical Minesweeping

Minesweeping is also performed influentially by towing devices that create false magnetic, acoustic or pressure signatures that are needed to explode influence mines.

Minehunting is the act of actively searching for mines primarily with the use of sonar devices. Minehunting is often used to search for specific mine types in a known mine danger area.

2. Parameters and Terminology

The AMCM mission has uniquely specialized terminology and operating parameters. For the purpose of clarity the following crew member functional definitions are used in the following sections of this thesis.

- Pilot: The person in the cockpit physically at the controls of the aircraft. His/Her tasks include but are not limited to maintaining aircraft velocity, heading and altitude parameters.
- Co-Pilot: The person in the cockpit not in physical controls of the aircraft. The copilot's responsibilities include but are not limited to navigating, operating avionics and performing necessary checklists. The co-pilot must also assume the pilots role in the event of an emergency.

- Aircrewmen: The enlisted personnel who perform all tasks in the aft portion of the helicopter necessary to complete a mission. Aircrewmen tasks include but are not limited to operating winches, hoists and cables that are attached to the "towed body".
- Load Stress: The stress (workload) imposed by increasing the number of channels (or sources) over which information is displayed to an observer.

3. Operations, Hardware and Processes

The following is a description of the Operations, Hardware and Processes that are unique to the AMCM community.

- Tension: The tensile force in the cable connecting the helicopter to the "towed body."
- Tow Boom: The tow boom is essentially the device which attaches the "towed body" cable to the MH-53E Fig (3). The tow boom is typically fixed to the aircraft in the cabin ceiling. The free end is allowed to pivot from the level position to the cabin floor as well as 30 deg left/right of centerline as illustrated in Fig(4).

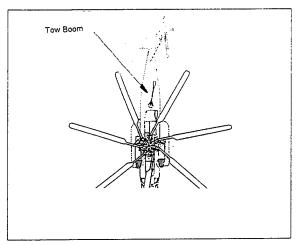


Figure 3 Tow Boom (Top View)

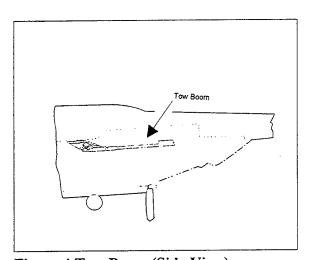


Figure 4 Tow Boom (Side View)

- Skew: The measured angle that the tow boom makes left or right of the aircraft's centerline during tow operations Fig (5). The skew must be monitored by the pilot during tow operation to ensure the "towed body" is properly positioned behind the aircraft.
- Drift: Very slow flight usually less than 15kts. Drifting is usually initiated from a hover and can occur in any direction (forward, backward or laterally).

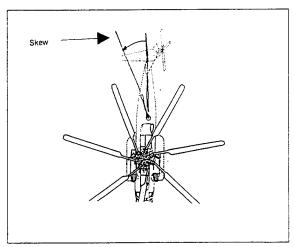


Figure 5 Skew

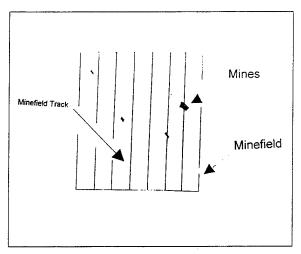


Figure 6 Minefield Track

Minefield Track: The intended flight path of the aircraft and the towed body during AMCM operations Fig(6). One of the primary tasks of the pilot during AMCM operations is to ensure that the aircraft flight path has a minimum deviation from the minefield track. Minefield tracks within a given minefield are discerned from one another by track numbers. • Yards to Remaining: The distance "a" remaining to the opposite end of the minefield Fig (7). Once the aircraft and "towed body" have transited this distance the process of turning the aircraft and "towed body" commences.

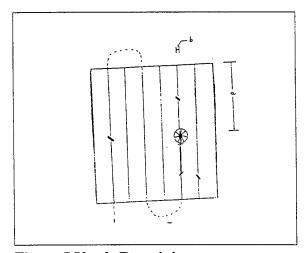


Figure 7 Yards Remaining

III. THE MH-53E AIRCRAFT AND MISSION SUBSYSTEM

A. THE MH-53E AIRCRAFT

Currently, the MH-53E Super Stallion is the U.S.

Navy's sole airborne mine countermeasures (AMCM) aircraft.

While other aircraft have been used in the past, the MH53E's superior size and power has made it the aircraft of

choice for the AMCM mission. The aircraft is manufactured by

Sikorsky Aircraft, a Division of United Technologies, located

in Stratford Connecticut Fig (8), (NAVAIR, 1993).

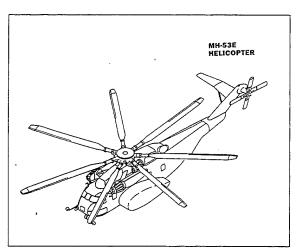


Figure 8 MH-53E

Some of the MH-53E design specifications and standard sea level performance characteristics are as follows (NAVAIR, 1993).

Geometric Parameters										
Length	99ft	½ in								
Height	28ft	4in								

Table (1) Geometric Parameters

Rotor Parameters	Main	Tail
Radius (ft)	39.500	10.00
Chord (ft)	2.440	1.28
Blade No	7.000	4.00
Solidity	0.163	0.138
Tip Speed (ft/s)	732	733
Airfoil Type	SC1095	NACA 0015

Table (2) Rotor Parameters

Aircraft Characteristics Engines: 3, T64-GE-416A @ 4380 SHP each Max. Cruise: 150 kts Max. Rate of Climb: 2,500 fpm (with 25,000 payload) Max Gross Wt : 69,750 lbs.

Table (3) Aircraft Characteristics

B. AMCM MISSION OUTLINE DESCRIPTION

A schematic of a typical AMCM mission is depicted on Fig (9). As noted above, primary mission phases are composed of Streaming, Towing and Recovery.

Stream Phase: The first phase of the AMCM mission involves the deployment of the "towed body" from the aircraft into the water before reaching the minefield. The process of streaming a "towed body" is physically accomplished by the actions of the aircrewmen.

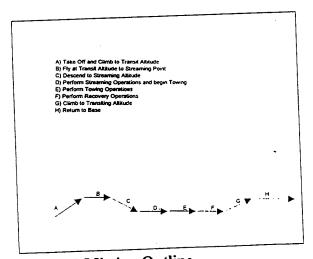


Figure 9 Mission Outline

The aircrewmen tasks primarily involve physically getting the "towed device" out of the aircraft into the water and back onto the aircraft. The aircrewmen essentially operate winches located in the aft cabin area which pay out cables attached to the "towed body". Only after the "towed body" enters the water can the AMCM operations of mine sweeping or minehunting begin. The aircrewmen perform the majority of their tasks in the aft cabin area as depicted in Fig (10). During the Streaming Phase the pilot's primary task is to fly the aircraft as dictated by NATOPS procedures, in such a way as to prevent oscillations of the "towed body." The primary mission objective during the stream phase is the minimization of the time required to stream the "towed body." Accomplishing this objective helps to maximize the time spent towing in the minefield.

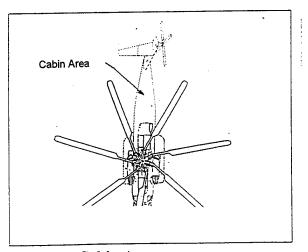


Figure 10 Cabin Area

Figure (11) depicts the aircraft position during streaming phase of an AMCM mission.

Towing Phase: As noted above the second phase of the mission is that in which the minehunting and/or minesweeping activity takes place. Figure (12) depicts the aircraft operating in the towing phase of an AMCM mission. During this phase the aircrewmen are primarily visually monitoring the "towed body". The primary objective during the towing portion of the mission is for the pilot to fly an accurate minefield track with the proper ground speed, tension and skew parameters.

Recovery Phase: The process of bringing aboard the towed body onboard the aircraft, ship, or beach line from which it was initially released. The recovery phase can essentially be thought of as the Streaming Phase in reverse.

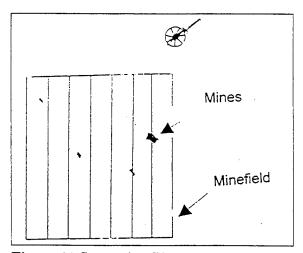


Figure 11 Streaming Phase

The pilot once again must fly the aircraft in such a way as to prevent oscillations while the aircrewmen winch in the "towed body". Figure (4) depicts the aircraft operating in the recovery phase of an AMCM mission.

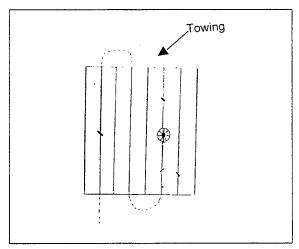


Figure 12 Towing Phase

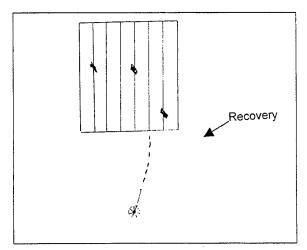


Figure 13 Recovery Phase

C. MISSION SUBSYSTEMS

The MH-53E accomplishes the minesweeping/mine hunting tasks by towing a device containing several subsystems, which are attached internally to the aft end of the helicopter, through the water. The following is a list of the subsystems used for AMCM minesweeping/minehunting tasks.

MK-103: Mechanical wire sweeping apparatus used for moored mines. The MK-103 Fig(13) accomplishes it's mission by cutting the submerged chain of a moored mine with explosive cutters. After the mooring chain is cut the mine rises to the surface of the water.

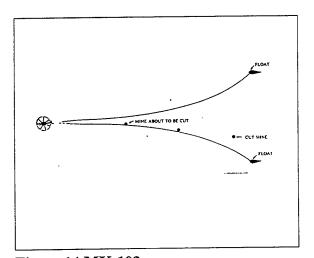


Figure 14 MK-103

- MK-104: Acoustic signal generating device which generates acoustic signals via the venturi effect. The MK-104 generates an acoustic signature that emulates a vessel moving through the water.
- MK-105: A Hydrofoil sled capable of producing a magnetic signature that emulates the magnetic signature of a vessel Fig (15). This system can be combined with the MK-104 to sweep magnetic/acoustic influence mines.

• AN/AQS-Q14: Down and side looking sonar device used for locating bottom and moored mines Fig (16).

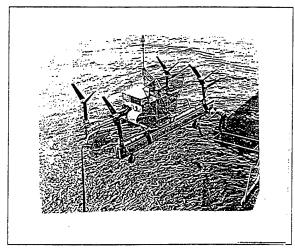


Figure 15 MK-105

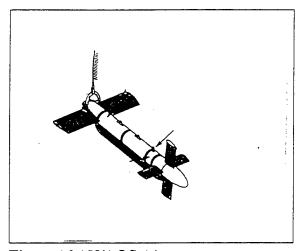


Figure 16 AN/AQS-14

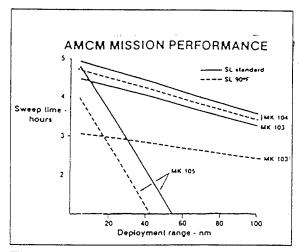


Figure 17 Mission Time

The AMCM subsystem time on station is unique for each particular device and varies linearly with the distance aircraft must travel to the minefield. Secondly, the time on station is distinctive because each "towed body" used has a unique approximate streaming time. The maximum minesweeping/minehunting time for most missions is approximately 4.7 hrs Fig (17). Finally, the AMCM towing capability is limited to a steady state tension of 25,000 lbs with a surge capacity of 30,000 lbs (NAVAIR,1993).

IV. MH-53E CREW MISSION REQUIREMENTS

As noted above the typical AMCM mission is executed in three phases Streaming, Towing and Recovery. As a reminder the person designated as the "pilot" is the person physically at the controls of the aircraft and the "copilot" is the person not in physical control of the aircraft. Lastly the "aircrewmen" are the enlisted personnel who perform all tasks in the aft portion of the helicopter necessary to complete a mission (i.e., sonar operation, device handling, etc.).

A. STREAM PHASE REQUIREMENTS

Stream Phase: The first phase of the mission involves the deployment of the "towed body" from the aircraft, ship, or beach line. Two of the primary objectives during the stream phase are the minimization of both the stream time and distance to the minefield Fig (11). Accomplishing these two objectives helps to maximize the time spent towing in the minefield. Fig (18) depicts the relative streaming position of a "towed body".

During the stream phase the pilot's tasks are concerned primarily with those things he must accomplish to provide the "best platform" to stream the "towed body."

The pilot is concerned with flying the correct altitude(s), heading(s) while maintaining the appropriate ground speed for streaming the "towed body". In addition to heading, altitude and ground speed cues, the pilot must also comply with the audio inputs from the enlisted aircrewmen directing the flight of the aircraft from it's aft section. This directing of the aircraft during the stream may require the pilot to drift the aircraft as a means of stabilizing the "towed body" as it is lowered from the aircraft to the water. Outside the cockpit the pilot must be cognizant of the wind direction and the status of the "towed body" which he can acquire from looking into the aft facing mirrors Fig (19).

The co-pilots' function is primarily to maintain oversight of the entire mission and to communicate the proper information to the pilot, aircrewmen and the air traffic controlling agencies at the appropriate time.

The co-pilot's additional responsibilities include advising the pilot, navigating, and controlling all avionics (e.g., checklists, channel switching, etc.)

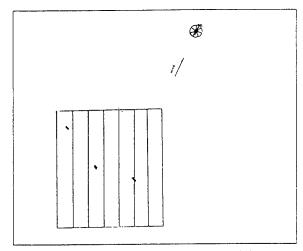


Figure 18 Relative Streaming Position

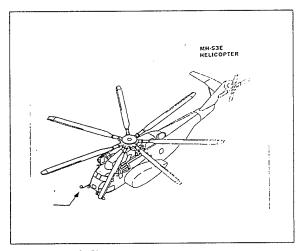


Figure 19 Mirrors

B. TOWING PHASE

Towing Phase: The second phase of the mission is that in which the active minehunting and/or minesweeping takes place. Fig (20) depicts the aircraft operating in the towing phase of a mission.

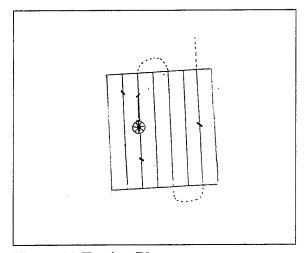


Figure 20 Towing Phase

During the towing phase of the mission the pilot's tasks include the acquisition of information he/she must have to properly tow the gear within the proper limits of ground speed, tension and skew. The pilot is also concerned with the aircraft's position in the minefield.

To facilitate accurate minesweeping/minehunting, the pilot must stay on the minefield track. Therefore, he must be cognizant of the aircraft position relative to the minefield track Fig (21). Near the successful completion of each minefield track, the pilot must be cognizant of the yards remaining on the present track, the direction to begin turning for the subsequent track, and the distance to the subsequent track.

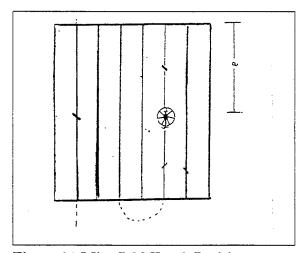


Figure 21 Minefield Track Position

Once the pilot has commenced the turn, he/she must have some indication of the rate of closure upon the successive track. This rate of closure information is critical to ensure towing accuracy and minimization of time outside the minefield.

During the towing phase of an AMCM mission the co-pilot requires that information necessary to advise the pilot of the minefield track prosecution sequence. This information includes the subsequent track number, relative distance, etc. Additionally, the copilot must continue to support the pilot by providing a back up scan to all cockpit instrumentation. Lastly, the co-pilot must maintain his communication with the air traffic controlling agencies and any conflicting shipping traffic, as necessary.

C. RECOVERY PHASE

Recovery Phase: The process of bringing aboard the towed body onboard the aircraft, ship, or beach line from which it was initially released.

During the recovery phase the flying aspects of the mission are much like the stream phase in that the pilot must provide the most steady aircraft platform possible to recover the "towed body." Similarly, the co-pilot's requirements mirror those required during the stream phase.

D. AMCM CREW COORDINATION

Any analysis of AMCM crewstations would not be complete without mentioning of crew coordination requirements inherent to the mission. The primary reasons for the amount of coordination necessary is largely due to crew size & mission procedures. Crew size includes 2 pilots and from 2 to 5 enlisted aircrewmen. The number of aircrewmen is a function of the type of "towed body" used. Plainly put, the pilots coordination between themselves and the crewmen is crucial to ensure mission safety.

V. CURRENT MH-53E COCKPIT

A. GENERAL DESCRIPTION OF INSTRUMENTATION

The current instrument panel illustrated in Fig (22) is labeled with numbers to identify which instruments are used by a pilot performing an AMCM mission. The following text provides a brief description of what type of information is furnished to the pilot by each individual instrument.

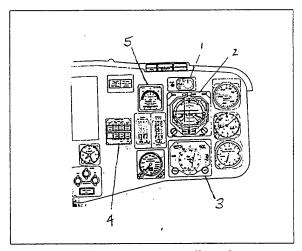


Figure 22 Pilot Instrument Panel

B. INDIVIDUAL INSTRUMENT DESCRIPTION

(1) Ground Speed Drift Angle Indicator (GSDA)
Drift Direction: An arrow indicates the direction in which the aircraft is moving (drift). Ground Speed is also numerically read out in kts Fig (23) (NAVAIR, 1993).

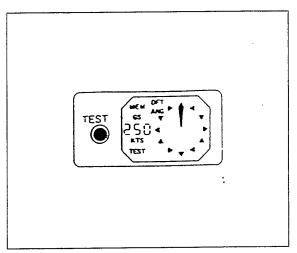


Figure 23 GSDA

(2) Attitude Directional Indicator(ADI)
The two ADI's installed on the instrument panel
visually indicate the helicopter's pitch, roll attitude,
turn rate, slip and navigational information Fig(). For
AMCM navigation the horizontal bar indicates the relative
ground speed and the vertical bar acts as a "steer to" bar
to remain on the minefield track .(NAVAIR, 1993) See Fig(24).

(3) Horizontal Situation Indicator (HSI)

The two HSIs installed on the instrument panel Fig(25), present a plan view of the navigational situation as if a pilot were looking down from above the helicopter. The instrument consists primarily of a rotating compass card, two bearing pointers, a heading indicator and course deviation indicator, (NAVAIR, 1993).

(4) Mode Selector Panel

Two mode selector panels marked MODE SEL are on the instrument panel Fig (26) to allow each pilot to select the source of heading and attitude reference to their respective ADI and HSI. These selector panels allow each pilot to select his/her preferred navigation source (i.e. VOR, TACAN, etc) (NAVAIR, 1993).

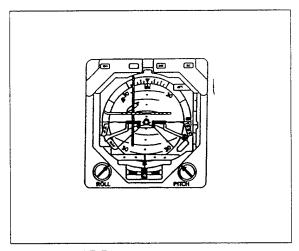


Figure 24 ADI

(5) Tension Skew Indicator (TSI)

During the towing portion of the mission this instrument provides the pilot with tow tension and skew angle information. The arced LCD tension scale in the upper center of the indicator provides two parallel displays.

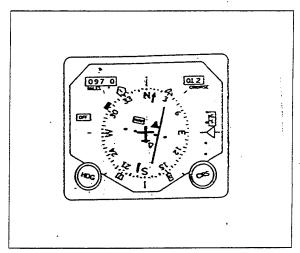


Figure 25 HSI

The upper scale is divided into segments from 0 to 40 x 1000 representing 1000 lbs each. The lower scale is divided into segments from 0 to 6 x 150 lbs each. The lower portion of the indicator provides a LCD scale showing skew angle from 12 degs left to 12 degs right. Below the scale indices at 1 deg increments are shown from 10 left to 10 right with 10, 0, and 10 marked Figures Fig (15) (NAVAIR, 1993).

(6) VO-30

The VO-30 located in the console serves to indicate relevant minefield distance (e.g. yards left of track, yards to go, etc). The VO-30 indicates numerically the aircraft distance from track and the distance to the end of the minefield. The VO-30 also indicates with bars the "fly to" direction. Additionally, the azimuth direction of the minefield and the track number is also indicated Fig(16) (NAVAIR, 1993).

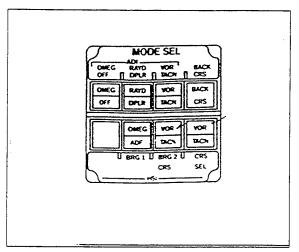


Figure 26 Mode Select Panel

(7) Global Positioning System (GPS)

The current global positioning system located in the cockpit console section provides precise navigation information in latitude/longitude. The current GPS system is capable of storing and providing the necessary navigation information to fly to preset navigation points Fig (17). The Global Positioning System provides the precise navigation necessary to conduct AMCM missions (NAVAIR, 1993).

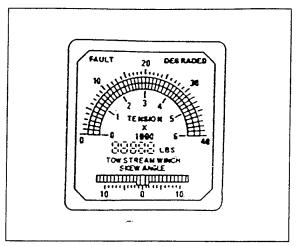


Figure 27 TSI

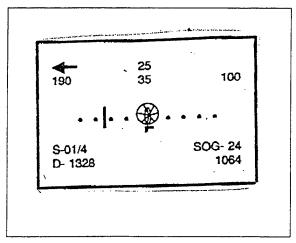


Figure 28 VO-30

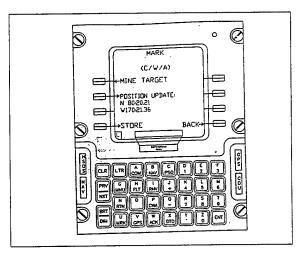


Figure 29 GPS

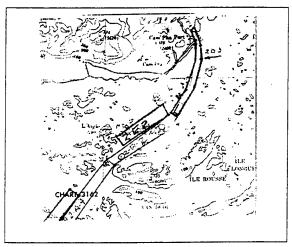


Figure 30 Navigation Charts

(8) Standard Nautical Charts
Included in the navigation charts for each AMCM mission is a standard nautical chart. The nautical chart includes the layout of the minefield azimuth and relative size. The chart also depicts pertinent hazards to navigation (e.g. reefs, buoys, and shipping channels, etc.) Fig(30) (NAVAIR, 1993).

C. MH-53E COCKPIT CHANGES

The MH-53E helicopter in its present state has a very high workload cockpit. Recently, a major design change to the MH-53E cockpit has been built and evaluated. This new "glass cockpit" known as the MH-53E Navigation/Communication System (NCS) is the most significant change to the Airborne Mine Countermeasures (AMCM) community since the use of global positioning system (GPS) navigation and should prove just as valuable.

The design changes in the MH-53E cockpit are primarily the replacement of navigation instruments that depict the helicopter's horizontal situation. This design change is unique in that this is the only aircraft in recent times where the cockpit has been vastly altered, but the mission and exterior aircraft design has remained unchanged .

VI. THE NEW MH-53E COCKPIT

The design changes in the MH-53E cockpit are primarily the replacement of navigation instruments that depict the helicopter's horizontal situation.

A. GENERAL DESCRIPTION OF INSTRUMENT CHANGES

The new MH-53E cockpit layout is shown in Fig(31). A contrast of the current cockpit and the new cockpit is featured in Fig (32). The old cockpit contains several instruments that were duplicated on both sides of the cockpit (i.e., airspeed indicator, attitude gyro, etc. The new cockpit design called for the removal of some dual instruments from both sides of the cockpit, and yet other instruments were only removed from a single side. Table (4) contains the list of instruments which were removed from the current cockpit.

Instrument	Left Side	Right Side	Both Sides
HSI			Removed
Mode Select Panel			Removed
TSI	Removed	Albert Califo Milita	
AMCM Caution Advisory Panel	Removed		

Table (4) Instruments Removed

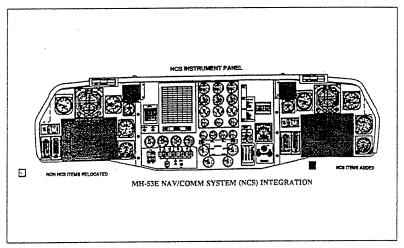


Figure 31 New Cockpit Layout

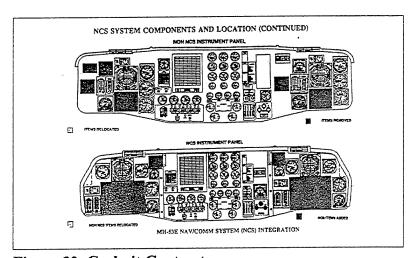


Figure 32 Cockpit Contrast

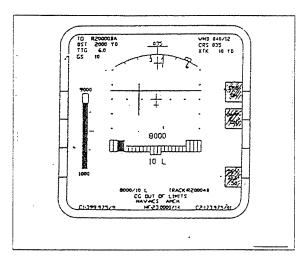


Figure 33 HSDS

The New cockpit primary horizontal reference instrument is the Horizontal Situation Display System (HSDS). The HSDS displays are located in position formerly occupied by the HSI's Fig(33).

B. HORIZONTAL SITUATION DISPLAY SYSTEM (HSDS)

Two HSDS displays are installed in the new instrument panel, one each side of the instrument panel. screens. The HSDS provide the pilots with aircraft attitude and horizontal directional information.

The HSDS provides the aircraft attitude and horizontal directional information to the pilots by allowing greater than 17 different screens to be displayed. The two screens used for AMCM operations of the HSDS are the TOW and MCM screens (NAVAIR, 1994).

1. HSDS TOW Screen

The HSDS TOW screen provides a moving map around an aircraft symbol in the center of the screen. For most AMCM operations the TOW screen will be primarily used by the copilot .The HSDS TOW screen displays also include symbols for minefield/towing operations. Figure (34) illustrates the moving map display with the current minefield and minefield track number in the middle of the screen. The current tension/skew text is the "1000/2 R" at the bottom center of the display in Fig (34). Adjacent to the tension skew text is the text indicating the current minefield track. The center of gravity text is illustrated directly below the tension/skew text.

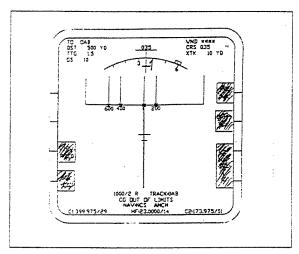


Figure 34 HSDS TOW Screen

The text indicating the navigation system and mission type are illustrated directly below the center of gravity text. The bottom line of text in Fig (34) designates the radio sources and frequencies. The Tow screen features in the top left of the screen include information such as the distance to the opposite end of the field, Distance To (DST); the time to travel to the end of the minefield, Time-To-Go (TTG); and the Groundspeed (GS). The text in the top right of the screen notes the direction and speed of the wind (WND), the Commanded/Desired course(CRS), and the Cross-Track Deviation (XTK). The "X" on each mine track indicates the track that is to be flown.

The darkend blocks on the right and left of the screen are the display control keys. The analysis of the display controls are beyond the scope of this thesis. Additionally, the minefield map will display a turn path computed by the NCS based upon device type and track separations (NAVAIR, 1994).

2. HSDS MCM Screen

The HSDS MCM (Mine Countermeasures) screen provides a compass rose, tension/skew and the minefield track number. For most towing operations, the MCM screen is primarily used by the pilot. These displays include a tension manometer, a skew bar, tension/skew text, cross track deviation bar and a ground speed deviation bar Fig(35).

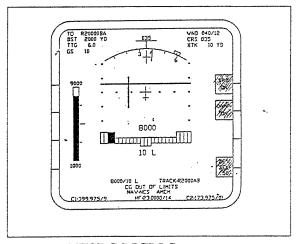


Figure 35 HSDS MCM Screen

The tension manometer is the vertical bar on the left of Fig (35). The tension manometer provides an illustration of the current tension along with the tension limits numerically indicated at the top/bottom of the bar.

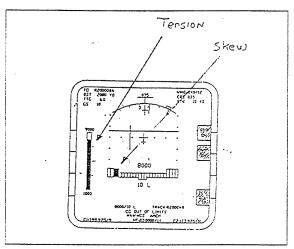


Figure 36 HSDS MCM Screen

The skew bar displays a scale graduated in degrees left to right, with a lighted block that moves left or right of center proportional to tow boom skew. Tension is displayed in the text above the skew bar to the nearest 100 lbs. Skew text is displayed below the skew bar indicating degrees left or right in Fig (37). The cross track deviation bar is centered about the aircraft symbol located in the center of the HSDS display.

The cross-track deviation bar represents the current track and deflects left or right. The bar is displaced proportional to the distance between the aircraft and the current minefield track as defined by the scale markings. The ground speed deviation bar and scale is centered about the aircraft symbol, at the center of the HSDS display. The ground speed deviation bar deflects forward or aft of the aircraft to indicate the aircraft groundspeed as slower (forward) of faster (aft) than that selected. The darkened blocks to the right are display control keys. The analysis of the display control keys are beyond the scope of this thesis.

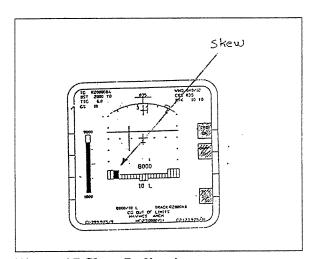


Figure 37 Skew Indication

VII. SIMPLIFIED AQS-Q14 MISSION ANALYSIS

In order to gain a more comprehensive view of how the current cockpit compares to the new NCS Glass cockpit, a partial Mission Task Analysis is illustrated in the following text. The upcoming mission task analysis serves to highlight some of the differences in the two cockpits as some of the primary mission tasks are performed.

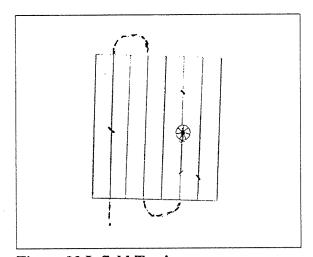


Figure 38 Infield Towing

This analysis should serve to highlight the noted cockpit differences in the performance of an AQS-Q14 mission. The phase of the mission chosen for comparison is the towing phase at two distinct parts of the phase, those being "in field towing" Fig (38) and "turning" Fig (39).

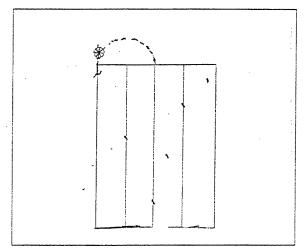


Figure 39 Turning

These two regimes of the towing phase were chosen because they are most able to suggest the differences in the required workload and situation awareness. The required workload and situation awareness aspects were chosen because of their significant importance as crew centered design criteria.

The AN/AQS-14, commonly referred to as the "Q14" is essentially a down/sidelooking sonar device. Fig(40) is an illustration of the approximate search path of a "Q14" conducting AMCM operations. Once deployed from the aft end of the aircraft the "Q14's" depth is subsequently controlled from a console by an enlisted aircrewman.

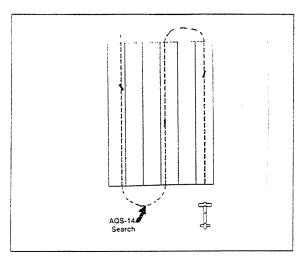


Figure 40 AN/AQS-14 Search Path

The subsystem can be used to locate both bottom and moored mines by the sonar reflections.

A. PILOT INFORMATIONAL REQUIREMENTS

For the towing environment during the ANS-Q14 mission, the primary information required by the pilot is as follows:

Required Pilot Information	
Tension in (lbs)	
Tension in a visual tape gauge format	
Skew in (degrees)	
Skew in a visual block reference gauge	
Ground Speed in (kts)	
Relative Ground Speed as a visual reference	
Distance to the end of the field in "yds to go"	

Table(5) Required Pilot Information

The following text serves to highlight the differences in the present cockpit displays vs. the NCS subsystem displays. The differences examined are those encountered by a pilot performing an AMCM mission. In order to compare the differences between the present cockpit displays vs. the NCS subsystem displays a uniform reference system was used. The uniform reference system assumes the distance to both the current cockpit display and the NCS system display are approximately equal for any particular pilot. The comparison makes references only to displacements in the plane of the cockpit instrument panel. The points about which the measurements were made were the respective centers of the two different primary attitude displays. Thus the reference point used for the NCS system is the center of the HSDS screen. Likewise, the reference point used for the present cockpit is the center of the ADI.

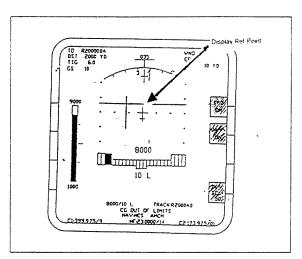


Figure 41 Reference Point

Both of these points correspond to the location where the cross track deviation bar and the visual speed reference bar cross as depicted in Fig (41).

Using the present system, the pilot must scan the following distances to acquire the necessary information listed in Table (6).

Information Required	Scan Distance
	<u>(in)</u>
Tension in (lbs)	5.25
Tension in tape gauge format	5.50
Skew in (degrees)	5.00
Skew as a visual block	5.25
reference	
Ground Speed in (kts)	3.75
Distance to the far end of	> 30.0
the present minefield	

Table(6) Information Required by Pilot

For the pilot using the NCS system the distances he must scan to get the same associated information are listed in Table(7).

Information Required	Scan Distance
	(in)
Tension in (lbs)	2.63
Tension in tape gauge format	1.13
Skew in (degrees)	2.13
Skew as a visual block	1.75
reference	
Ground Speed in (kts)	2.50
Distance to the far end of	2.63
the present minefield	<u> </u>

Table (7) Information Required by Pilot

B. CO- PILOT'S INFORMATION REQUIREMENTS

The information required by the co-pilot is quite different than that required by the pilot because of the greater variety of tasks performed by the co-pilot. The co-pilot's tasks include but are not limited to navigating, tuning avionics, performing checklists ,etc. In regards to the co-pilots information requirements a comparison of the NCS cockpit vs present cockpit was made. However, since the co-pilot's required information is in proximities other than instrument panel, a scan distance comparison was not made. For the co-pilot's information requirements the only comparison made involved instrument proximity. For a co-pilot in the present cockpit the instrument locations are listed in Table (8).

Information Required	Instrument Location	
Aircrafts minefield position	VO- 30 (center console)	
Distance to next minefield	VO- 30 (center console)	
track		
Direction of turn	VO- 30 (center console)	
Time to turn cue	VO- 30 (center console)	

Table(8) Instrument Proximity

For a co-pilot in the NCS cockpit the instrument locations are listed in Table (9) .

Information Required	Instrument Location	
Aircraft minefield position	HSDS Front Inst Panel	
Distance to next minefield	HSDS Front Inst Panel	
track		
Direction of turn	HSDS Front Inst Panel	
Time to turn cue	HSDS Front Inst Panel	

Table (9) Instrument Proximity

It should be noted that the flat position of the VO-30 screen often requires the co-pilot to bend over it to acquire the necessary information. The co-pilot positioning himself in this manner is often due to glare and the VO-30's small screen size.

Turning Comparison

The following comparison of the NCS system vs. the present cockpit is done to highlight how the differences in the cockpit configurations could effect mission effectiveness. The "turn" portion of the mission was selected for comparison because this portion of the mission is directly related to mission effectiveness.

The effectiveness of a AMCM mission can be measured by the Operational Tow Time. The Operational Tow time is a good measure of effectiveness because it measures the amount of time spent in the minefield performing tow operations.

The Operational Tow Time commonly referred to as OPTOW can be thought of using the following mathematical relations:

T = TOTAL TOW TIME

ST = STREAM TIME

TU = TIME IN TURNS

AV = TIME IN AVOIDANCE TURNS

$$OPTOW = (T) - (ST) - (TU) - (AV)$$
.

Thus, effectiveness of Aircrew Centered System Design during this part of the mission is paramount because the effectiveness of the turn is substantially dependent on the crews situational awareness.

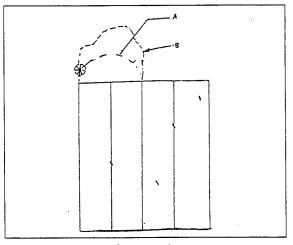


Figure 42 Turn Comparison

The significant differences in the cockpit displays are that in the new cockpit:

- 1) The co-pilot and pilot can see the complete minefield from an moving map perspective.
- 2) The pilot does not have to rely on the co-pilot to inform him/her of when to start turning.

As an illustration, Fig (42) depicts the differences between a pilot making an efficient turn, (A) vice an non-efficient turn, (B). It must be kept in mind that the towing speeds are very low during operations.

C. TURNING COMPARISON

In order to compare the cockpit displays during a turn it is first necessary to understand the type of information that is required of the pilots. For the pilot the required information is as follows.

Required Pilot Information	
Tension in (lbs)	
Tension in a visual tape gauge format	
Skew in (degrees)	
Skew in a visual block reference gauge	
Ground Speed in (kts)	
Distance to the end of the minefield	
Time to Start turn Cue	

Table (10) Required Pilot Information

For the pilot using the present cockpit the required information and associated scan distances are listed in Table (11). Similarly for the pilot using the NCS cockpit the required information and associated scan distances are listed in Table (12)

Required Pilot Information	<u>Scan Distance</u> (in)
Tension in (lbs)	5.25
Tension in a visual tape gauge format	5.50
Skew in (degrees)	5.00
Skew in a visual block reference	5.25
gauge	
Ground Speed in (kts)	3.75
Distance to the end of the minefield	Not Available
Time to Start turn Cue	Not Available

Table (11) Scan Distance

Required Pilot Information	<u>Scan Distance</u> (in)
Tension in (lbs)	2.63
Tension in a visual tape gauge format	1.13
Skew in (degrees)	2.13
Skew in a visual block reference	1.75
gauge	
Ground Speed in (kts)	2.63
Distance to the end of the minefield	2.63
Time to Start turn Cue	2.63

Table (12) Scan Distance

VIII. OPERATION EVALUATION (OPEVAL)

A. OPERATION EVALUATION BACKGROUND

The NCS system has undergone a thorough fleet evaluation otherwise know as an "OPEVAL." The OPEVAL was conducted in 3 March to 27 April of 1995 under the supervision of Air Test and Evaluation Squadron One (VX-1) of Patuxent River Maryland.

The scope of the OPEVAL was to take the NCS mission system currently loaded onto a fleet standard MH-53E helicopter and let fleet pilots fly and evaluate the system in their training environment. The fleet evaluation pilots selected were five MH-53E pilots in number with varying degrees of experience, stationed at Helmineron Fourteen (HM-14) located at NAS Norfolk , Virginia. Prior to the evaluation flights, the pilots received ground school instruction by the NCS system manufacturer, EER Systems Inc. After the ground school training, the pilots received familiarization flights with Squadron ONE's MH-53E NCS system evaluation pilot. Once familiar with the NCS system operation , the HM-14 pilots conducted seven AMCM training sorties after which the pilots completed an AMCM Mission Evaluation Narrative (NAVAIR, 1995).

The AMCM Mission Narrative authored by (VX-1) contained nine questions in which the pilots were asked to describe system performance in the AMCM environment. An example of the AMCM narrative is contained in Figs (43).

B. OPERATION EVALUATION RESULTS CLASSIFICATION

The author evaluated the data from the narratives in a concise manner, the data was separated into four categories: Orientation & Situational Awareness, Marking Procedures, Navigation & Steering, Weight and Balance Calculations. Since the nature of the narrative data was subjective and represented at best ordinal data only three levels of measurement was used to quantify the data. The author utilized three levels of measurement which were Favorable, Neutral and Unfavorable (Petho, 1995).

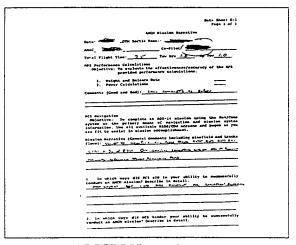


Figure 43 AMCM Narrative

C. RESULTS

The results of the AMCM narrative summaries are indicated in Fig (44). From the summary data it is very apparent that pilot opinions of the cockpit design changes were primarily favorable or unfavorable. The aspects of the design changes that yielded the most favorable results were the Orientation/Situation Awareness & Navigation /Steering. The most unfavorable aspect was the marking procedures. The marking procedures received unfavorable opinions because they required the co-pilot to input: time, skew angle, etc. into the GPS interface in order complete the marking procedures.

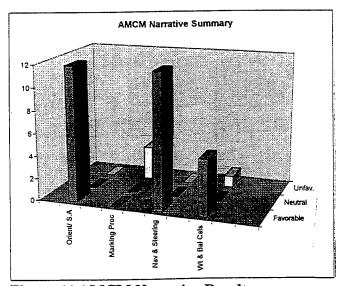


Figure 44 AMCM Narrative Results

IX. IMPACT OF AIRCREW REQUIREMENTS ON AIRCRAFT DESIGN

The subject matter explored in this part of the thesis involves some observations of possible design changes that could be undertaken in future AMCM aircraft.

A. ROTOR DOWNWASH REDUCTION

Rotor downwash is a phenomena of which AMCM aircrews must constantly be aware, because of it's negative effects on mission performance. Rotor downwash is the downward airflow produced by the rotating main rotor of a helicopter as depicted in Fig (45).



Figure 45 Rotor Downwash

Rotor downwash can have a negative impact on AMCM mission performance by causing water mist to obstruct the aircrew's vision during an over water hover or by oscillating the AMCM "towed body" during streaming/recovering operations. When vision obstruction problems become apparent this often forces the crew to hover at a higher altitude than if no misting had occurred. The problems of oscillating "towed bodies" are often rectified by the pilot adjusting the aircraft drift speed/direction. Rotor downwash is the primary reason the MH-53E helicopter must hover over water at approx 50+ ft vice 35+ ft accomplished by smaller aircraft like the SH-3, SH-60, etc. This required hover altitude due to rotor downwash misting impacts the AMCM mission by sometimes increasing the amount of time required to stream/recover a "towed body".Lastly, one of the reasons MH-53E aircrews are not authorized to hoist individuals during an over water rescue is because of the vigorous rotor downwash induced waves to which a person would be subjected.

The phenomena of rotor downwash is primarily a function of the helicopter's Disk Loading (DL) (Prouty, 1990).

Mathematically, Disk Loading is equal to rotor thrust divided by the rotor disk area as suggested in equation (1).

DL = Thrust/Rotor Disk Area (1)

For an helicopter in a hover, this thrust is equal to the Gross Weight (GW). The relationship between the air downflow velocity and the disk loading at sea level is expressed by equation (2).

$$v = \sqrt{\frac{(DL)}{2\rho}} \tag{2}$$

Based on equation (2), Fig (46) depicts the relationship between air downflow velocity and the helicopter's rotor radius for various aircraft gross weights. The design parameters by which the engineer can vary the aircraft's rotor downwash are primarily the aircraft's Gross Weight (GW), and rotor blade length (R). The aircrafts' weight is dependent upon its component weights while the rotor disk area is a function of the rotor radius.

Equation (1) can thus be rewritten in terms of these parameters as equation (3) (Prouty, 1990).

$$GW = \pi R^2 DL \tag{3}$$

Based on equation (3), Fig (47) depicts the relationship between air downflow velocity and the helicopter's rotor radius for various aircraft gross weights. Therefore, changes in a helicopter's gross weight or rotor blade length will have an effect on the helicopter's downwash velocity and possibly the aircrew operating the helicopter. Lowering an aircraft's disk loading may allow an aircrew to clearly see at altitudes lower than previously possible.

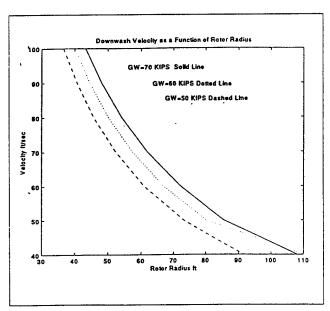


Figure 46 Downwash vs Rotor Radius

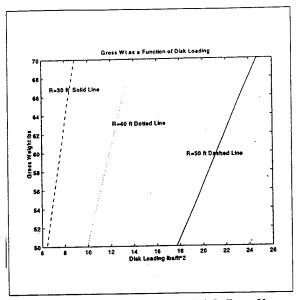


Figure 47 Gross Weight vs Disk Loading

B. COEFFICIENT OF DRAG REDUCTION

In an effort to explore the manner in which the space requirements of the aircrew are related to total aircraft drag, a brief study was conducted in which the aircraft fuselage size was related to the coefficient of drag. If the aircraft fuselage diameter is narrowed by 10% the aircraft total drag could be reduced by a minimum 2.5%.

This reduction in drag has a potential to significantly affect the mission by decreasing the required thrust. The workspace of the aircrew is best illustrated by a photograph as illustrated in Fig (48)which is taken inside the MH-53E aircraft fuselage looking aft.

As depicted during an AMCM mission, the aircrewmen workspace is very limited. During an AMCM mission the aft end of the MH-53E contains winches, consoles and the "towed body". One requirement that cannot be compromised is the requirement that the aircrew must have sufficient workspace within the aircraft in which to escape, in the event of an aircraft ditching. Thus, a reduction in drag as a result of a narrower fuselage would impact the aircrew by making for a more confined cabin.

A smaller cabin area would necessitate a requirement for smaller winches in order to accommodate the crew ditching requirements. The potential use of smaller winches could only happen if the "towed bodies" are re-designed to require less tension.



Figure 48 Aircrew Workspace

Thus, from this aspect the needs of the crew could have an indirect impact on the decision to make the aircraft fuselage smaller or in the re-design of the "towed body."

X. CONCLUSION

It is suggested that the proposed changes in the MH-53E instrument panel should reduce the pilot load stress (Petho,1992) by

- 1) Reducing the number of instruments the pilot must scan to perform the AMCM mission.
- 2) Reducing the distance the pilot must scan to perform the AMCM mission.

It is suggested that Situation Awareness and Navigation/Steering aspects of the new cockpit were the most favorable results because it provided the pilots with a top view of where the aircraft was with respect to the inside and perimeter of the minefield. It is also suggested that the positioning and screen size of the VO-30 in the present cockpit may pose a hindrance to the pilot attempting to acquire necessary information.

Contrarily, the HSDS TOW screen may significantly improve the ability of the pilots to acquire information by

1) Positioning the screen on the primary instrument panel where it's proximity for glare is reduced. The positioning of the screen additionally enables the pilot to obtain the moving map thereby enabling him to judge his rate of turn.

- 2) The HSDS TOW screen provides an complete overview of the mine field on a much larger screen than does the VO-30 .The larger screen size of 6 X 6 inches enables the information to be scanned much more easily than on the $4\ X\ 5$ inch VO-30 screen .
- 3) The automatic turn indicator provides takes the guesswork out when the pilot should start the AMCM turn.

The procedures used in the marking of a mine-like contact proved to be the only source of unfavorable comments. Three out of the fourteen narratives state that the marking of a "mine-like" contact was too laborious because it prompted the pilot to input the location of the "mine-like" object. This noting of the "mine like" object is not required because the aircrewman performing the Recorder tasks are required to record this information as per NATOPS procedures. Thus, the negative comments concerning the "marking" procedures should not significantly impact the NCS system performance.

XI. RECOMMENDATIONS

It is recommended that more detailed explorations be made as to how aircrew design requirements impact weight and or drag of helicopters. Secondly, it is recommended that a study be conducted to see if the new NCS system has positive impact on operational tow time.

APPENDIX



Data Sheet E-3 Page 1 of 3

Date: 17APR DTM Sortie Name: 17APRQ1401
AHAC Co-Pilot
Total Flight Time: 1.8 Tow Hrs Op Tow
MPS Performance Calculations Objective: To evaluate the effectiveness/accuracy of the MPS provided performance calculations.
1. Weight and Balance Data 2. Power Calculations
Comments (Good and Bad): No PROBLEM;
NCS Navigation Objective: To complete an AQS-14 mission using the Nav/Comm system as the primary means of navigation and mission system information. Use all available HSDS/CDU screens and pages as you see fit to assist in mission accomplishment. Mission Narrative (General comments including minefield and tracks
Chilized MPS Stream point. Completed several
tracks. Mission about due to 0-14 Props.
RTB vio fly plan rouk.
1. In which ways did NCS aid in your ability to successfully conduct an AMCM mission? Describe in detail. Vastly increased stational awareness. Fellowing
The strain point, we manually sequenched to first track For some reason, we lost the PSED (it sequenched to last track)
For some reason, we lost the PSED (it segrenched to last track)
Tow screen still allowed me to navigate
2. In which ways did NCS hinder your ability to successfully conduct an AMCM mission? Describe in detail. Publem with towed budy pusition.

Did the NCS provide adequate steering cues and the ability to navigate precisely in the minefield? NES NO If no, why? YES Looking over a copility Tow Screen makes acceptant and easy.
4. Which NCS functions, if any, were particularly difficult, time consuming, or required too much focus in the cockpit?
MARK, - instally takes a lot of time to infut all
the variable)
5. Were there any failures or difficulties with the NCS encountered during the flight? YES NO Describe:
After munually sequencing once, NIS sequences
After munually sequencing once, NIS sequences to last truck of minetical
Did you see any abnormal CWA's?
No
7. Was critical mission information readily available?
705
8. Did you deviate from your MPS created flight plan? If yes, why and how? We flew our route axactly as flight plans

9. Do you fores integration?	ee any	AMCM	tactical	changes	as	а	result	of	NCS
					· · · · · · · · · · · · · · · · · · ·				

Date: ZbAPR DTM Sortie Name: ZbAPRQ1408
AHAC Co-Pilot
Total-Flight Time: 2.8 Tow Hrs 1.8 Op Tow 1,5 ?
MPS Performance Calculations Objective: To evaluate the effectiveness/accuracy of the MPS provided performance calculations.
1. Weight and Balance Data 2. Power Calculations
Comments (Good and Bad):
Good
NCS Navigation Objective: To complete an AQS-14 mission using the Nav/Comm system as the primary means of navigation and mission system information. Use all available HSDS/CDU screens and pages as you see fit to assist in mission accomplishment. Mission Narrative (General comments including minefield and tracks
Mission Narrative (General comments including minefield and tracks flown): Another great flight of the NCS a.k. Standard munifield
(1) (4) (4) (4) (4) (4) (4) (4) (4) (4) (4
Splan, approx & perior complete. works sure received to Kew w/
Stream, approx & tracks complete. Winds were heavier reguling use of skewhold. 1st time AHAC manually controlled skew w/ MS system. 9kew Indicator works week, better than current 75.7
· /9
1. In which ways did NCS aid in your ability to successfully conduct an AMCM mission? Describe in detail. Skewindwater is specied to current TSI unit. Display is bigger!
more sold the and right there in front of you, The colored sections - fellow at
8° relat 12° - 15 actually give useful. If for example, so hatrone
8°, red at 12° - 15 actually give esel. If for example, so hatrond anareness lapses - that thing training yellow is easily seen from the
peripher 2. In which ways did NCS hinder your ability to successfully conduct an AMCM mission? Describe in detail. High for not even thoughton Not necessarily a hindrance and before this flight not even thoughton.
Not necessarily a historie and before this flight not even thought of
I redited on this flight tolair w/ the HATIC in the less sent a compiler the
the right seat, and withe copilot towing (ie: up on the MCM page) and the Hotte
on the you pages that ANAC MS NO reference to skew, along Them the
digital numbers on the bottom of the page (10: 84, 32, etc), or looking

00 / 100101	min the operal environment (using skewhold mostly, and es know how to tow), this could be a problem to the fleet
navigate	the NCS provide adequate steering cues and the ability to precisely in the minefield? If no, why?
consumir	th NCS functions, if any, were particularly difficult, timeng, or required too much focus in the cockpit? Governors Too much button masking uf formula to the cockpits of the context of the cockpits and the cockpits.
one fin	of Contacts requires Too much button mashing up for sole of Later link between The WC'D skew / heading / console
encounte	benefit in accurate Target relocation. There is too much on hecording miscliff contacts, either pilot or with the ere there and failures or difficulties with the Not ered during the flight? YES NO Describe:
6. Did	you see any abnormal CWA's?
No	
No 7. Was	critical mission information readily available?
<u> </u>	critical mission information readily available? you deviate from your MPS created flight plan? If yes, where the state of

where you have the co-pilot towing and he "loss the bubble" and skew gets way out, requiring the AHAC to take controls and Rectify the sitration. In the current configuration the AHAC and co-pilot both have TSI's, allowing the AHAC to mon, wor/ 8ch the co-pilot's skew, with the NCS configuration The AHAC STHONG on the Left seat would have to: A) convert a distribut output, like "10L" Into an "analog" interpretation and (1e: "My skew is way out to the lost, I need right pedal.", B) look cross carpit to get the proper protine and then make the correct assistment; c) switch his 450s from # Tow to MCM; or D) use an aircrewmen call from the back to make a determination and then correction. All of the above require way for much time and/or "brains" from the Attar. Correcting skew in Such a scenarto Is an mostrative reaction who any interpretation of time. There are different methods used by different pilots. I use the "Step on the zero" method - 10: w) The current 75% thue is a "O" indicated at the center of skew. If the skow's way out to the left the "0" is to the right of that and "stepping on the zero" (iciright pedal) is the correct mat regulard. This might sound like it require some shorght, but in practice 'it is a very guick and institutive reaction. With any of A through D abone, The time regisard world result in arresaft damage andlor personal injury.

Possible solutions might be to: A) have the AHAC occasstonally Switch screens to monitur skew (or look crow cockpit) This is not a satisfactory areans though, because the now - Clying pilot needs to be up on the a good turn. Out of the field and making a turn is also themost likely place for a pilot to got into trouble all skew.

B) MODIEG THE HSDS TOW PAGE to have a skew indicator Probably Genost exponsive/least feasible option, but the one I work Itke to see.

c) Make a procedural /technique modification, where the (next page,

After analyzes who he is lying with If it is someone he trusts not to pot him in a bad situation (ie: a good tower) then to seating arrangements don't mater. If the consider is someone finion of little experience, make him sit in the left seat and have the AHAC sit in the right seat where he can adequately monitor / backup the co-pilots skew by scanning the redundant. TSI on the right side.

an insue that reeds to be addressed.

9. Do you foresee any AMCM tactical changes as a result of NCS integration?

Possibly having AITAC GST right sent ul junios co-putato as mentionned before.

Date: ZLAPR DEM Sortie Name: ZLAPRQ1409
AHAC Co-Pilot
Total Flight Time: 7-5 Tow Hrs /, 8 Op Tow /, 0
MPS Performance Calculations Objective: To evaluate the effectiveness/accuracy of the MPS provided performance calculations.
1. Weight and Balance Data 2. Power Calculations
Comments (Good and Bad):
NCS Navigation Objective: To complete an AQS-14 mission using the Nav/Comm system as the primary means of navigation and mission system information. Use all available HSDS/CDU screens and pages as you see fit to assist in mission accomplishment.
Mission Narrative (General comments including minefield and tracks flown): Flouress Albs-14 The UF HIT SIMPLETED
R575, RISI, R 697, RZII, L575° R333 PAZINLLY
1. In which ways did NCS aid in your ability to successfully conduct an AMCM mission? Describe in detail.
ORKITATION WAS OUTSTANDING
2. In which ways did NCS hinder your ability to successfully conduct an AMCM mission? Describe in detail.
MARK PROCEOURE IS TOO LABORIOUS IF YOU INCLUDE
WHET SCREGO PERCENTAGES.

Data sheet E-3

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9. inte	Do you gratior	fores	ee any i	AMCM tact	ical ch	anges as	a result	of NCS
			NO	IFY	DU D	SCOUNT	TRACK	L AND
56	ECTUR	_ Da	TO DR	D ROME	WCIAI	MAF		

Date 2914195 DTM Sortie Name: 29414895 II
AHAC CO-Pilot LT
Total Flight Time: 30 Tow Hrs/ Sop Tow 1
MPS Performance Calculations Objective: To evaluate the effectiveness/accuracy of the MPS provided performance calculations.
1. Weight and Balance Data 2. Power Calculations 4.0
comments (Good and Bad): HOGS E Dad Engine Wast dispet ciffered from (and color lated value by 2 K# (was >Kless 4hen naturally salutates). Pur RECID / Avanable very identical to manually (also late. Don't inderstand disciplining,
NCS Navigation Objective: To complete an AQS-14 mission using the Nav/Comm system as the primary means of navigation and mission system information. Use all available HSDS/CDU screens and pages as you see fit to assist in mission accomplishment.
Mission Narrative (General comments including minefield and tracks flown): Flex dragon pad, XEay, & Moan foint w/ Completion of Hospitalism PSEC, followed by Same Norte fresh (Lid SWintry) Strational Revenues on Steen Him recovery (and accuracy) will depoin Excellent 5A!
1. In which ways did NCS aid in your ability to successfully conduct an AMCM mission? Describe in detail. Father is not of the full manuscrip for rest track Assellent SA in fact of field. Seriously increases OPTOW Sue to lidution of out The field managements
2. In which ways did NCS hinder your ability to successfully conduct an AMCM mission? Describe in detail. Dopper placement to thome forme gettern und to a Cross continue loshing not as flacifile as I thought it would be. Not a lighter thing just means that the copilat still plays are used in all its job, smith

	Data sheet E-3 Page 2 of 3
3. Did the NCS provide adequate steering coavigate precisely in the minefield? "YES NO If no, why?	cues and the ability to
	······································
. Which NCS functions, if any, were partice consuming, or required too much focus in the	
Vene of rate = == 51 important to co. to have lite pilots engaged enside	the cockpit sime Hanners
. Were there any failures or diffic encountered during the flight? YES NO De	culties with the NCS
History (probably operator one) at to Seguen & to mer al generic & then be eron, could not get forto seguence to "To	out to Ato Farance -engage "on toward
Did you see any abnormal CWA's?	7
Pilet ADIMCP	
. Was critical mission information readil	y available?
405-	
B. Did you deviate from your MPS created flown how? yis - did not stick sightly to Stocken portrack - Carrie in Surentry V	
El Track - Come in Skilentey V	ne / Ruy

.

9. Do you foresee any AMCM tactical changes as a result of NCS integration?

NCS will greatly encrease OPTOW-Saving Feel, time on Station, & Milliasing Mission "efficiency"

A great system. Very lice.

Passible Nice to see simproveness— a scale multimate between 4 a/6 of a "332" scale

The Hillzatron of the existing cyclic button "blank"

to allow the flying pilot to either Scale up a down

of to suith between the MCM etan pages

without having to remove highly from cyclic

(or to scan cursicochpit) would be a

Very live feature Don't know it it would be

a too substantial modification or not

AHAC		<u>-)</u>	Co-Pilot			
Total F	light Time:	7.8	_ Ťow Hrs_/	. 8	Op Tow_	1.4
MPS Peri Obje	formance Ca ctive: To e prov	valuate the	effectivene mance calcul	ss/accum ations.	racy of t	the MPS
1.	Weight an Power Cal	d Balance Da culations	ata		<u> </u>	
Comment	s (Good and	1 Bad): <u>Ac</u>	icrate -			
system informa see fit	tive: To as the pr tion. Use a to assist	imary means all availabl in mission (General Col	AQS-14 mis of navigate HSDS/CDU saccomplishme mments included the light Control of the light	nt.	and page	n system es as you nd tracks
		250/100.		L		
conduct	an AMCM m	ission? Desc	aid in your ríbe in deta ,-optow + M	111.	a trance	(Aussier
conduct	an AMCM m	ission? Desc	ribe in deta	111.	a trance	(Aussier
conduct flective Greatly Vie pe	an AMCM mild united in lanced in pheally (Skew Indic 2 Skew Indic 2 Skew indic 2 Skew indic	iribe in deta ,-optow tive ator right hold Kirted a licator & Ma	Me. Sit in from Mairie e yellor	tationa tog yer al time light.	(Aumin (pasy t p, but bic : t was (
conduct fective 6 ceitly (1) se pe Tyle fi 2. In	an AMCM mid united in lanced in the cally () which ways	Jesion? Description? Description? Skew India April - 5 Kew Le 5 Kew in a a did NCS 1	aid in your pribe in deta ater right hold Kirthold Kirthold Kirthold with the control of the co	in from Mairie Moliver Moli	to yet to suc	(Awaren (pasy t p, fut bec : t was (cessfully

Data sheet E-3 Page 2 of 3 9. Do you foresee any AMCM tactical changes as a result of NCS integration?

System provides for better mission of finishing based on increased op tow frought on by better strational curareners out of the field iffeitively reducing time "nowted" out of the field.

Perult is less five used priates time on station.

O. e. all - System is quantem lap improvenistores current
optim. It would be nice to have the ability ViA cyclic
better, to switch from Tow screen to MCM series w/c
removing hands from workers burnetly system is highly accurate
with a good co-pilat, but filet of ill relies heavily on copilat
to a good efficient turn. It would be a very nice improvement
to give the pilat the ability to monitors his own tron w/ the
purs of a cyclic butter.

Excellent flight

A STATE OF THE STA

Date: 29MAR95 DTM Sortie Name: 29MAR95
AHAC Co-Pilot
Total Flight Time: 3.0 Tow Hrs 1.8 Op Tow 1.2
MPS Performance Calculations Objective: To evaluate the effectiveness/accuracy of the MPS provided performance calculations.
1. Weight and Balance Data 2. Power Calculations
Comments (Good and Bad): FX(FLIFNG FL)(A)
NCS Navigation Objective: To complete an AQS-14 mission using the Nav/Comm system as the primary means of navigation and mission system information. Use all available HSDS/CDU screens and pages as you see fit to assist in mission accomplishment.
Mission Narrative (General comments including minefield and tracks flown): Executal MINERIA ORIGINATION AND FORTHUR STURING
ALMPENES EXSIER TO USE & EXPOSERE INCREASES, LEVER AT A LOSS
DURING LAUIGATING THIROUGH AND AROUND THE MINEFIELD
1. In which ways did NCS aid in your ability to successfully conduct an AMCM mission? Describe in detail.
MCM / TOW SCREENS MUCH EASIER TO OBSERVE AND ASSIMILATE DATA
HALF TO RELY ON COPILOT FOR STEEPAGE. TOO DIFFICULT JO LOOK
OUGR TO COPILOTS HOSS WHEN UNDER TOWN.
2. In which ways did NCS hinder your ability to successfully conduct an AMCM mission? Describe in detail. STREAM POINT THAT MPS PRIS IS TOO PESTILITIES.

	ge 2 of 3
3. Did the NCS provide adequate steering cues and the all navigate precisely in the minefield? YES NO If no, why?	oility to
4. Which NCS functions, if any, were particularly difficuous consuming, or required too much focus in the cockpit?	ult, time
EUROUTE SEGMENT TO SIREAY POINT	
5. Were there any failures or difficulties with encountered during the flight? YES NO Describe:	the NCS
6. Did you see any abnormal CWA's?	
7. Was critical mission information readily available?	
¥5.	
8. Did you deviate from your MPS created flight plan? If and how? YES, DEALT FILST TORKE OUT TO TIME WELFT.	DELETED
LAST TRACK AND MANUAL SEO TO NEXT LASTPORT. DETER	2 TH/NT
SYSTEM WOULDN'T AUTO SEP. THROUGH REST OF FLIGHT ,	PLAN

9.	Do you	foresee	any AMC	1 tactical	l change	s as a	result	of NCS
inte	gration?	DE LIA	F TO HA	UE JIMP	SENT O	GBSFRUHR	441	ZIE
		DL NIC	70 17.1					
TIME	<u> </u>							

Date: 24App = DTM Sortie Name:
AHAC CO-Pilot
Total Flight Time: 1,5 Tow Hrs 1,3 Op Tow . 5
MPS Performance Calculations Objective: To evaluate the effectiveness/accuracy of the MPS provided performance calculations.
1. Weight and Balance Data 2. Power Calculations
Comments (Good and Bad): So No Problems
NCS Navigation Objective: To complete an AQS-14 mission using the Nav/Comm system as the primary means of navigation and mission system information. Use all available HSDS/CDU screens and pages as you see fit to assist in mission accomplishment.
Mission Narrative (General comments including minefield and tracks flown): Departed X-RAY. FLEW MPS route to W-508
Slight deviation from Stream point. Manually sequenced interfect. Completed to tracks. Recovered due to Wx.
Flew back using MBS route. HAR to deviate due to
WX.
1. In which ways did NCS aid in your ability to successfully conduct an AMCM mission? Describe in detail.
Great Help! Marginal Wx about 600-1 at times
NCS provided outstanding Situational awareness.
2. In which ways did NCS hinder your ability to successfully conduct an AMCM mission? Describe in detail. The did not hinder at all.

Data sheet E-3

Date: 29 MAR95 DTM Sortie Name: 29MAR954
AHAC LT Co-Pilot LT
Total Flight Time: 2.5 Tow Hrs 1.6 Op Tow 5.9
MPS Performance Calculations Objective: To evaluate the effectiveness/accuracy of the MPS provided performance calculations.
1. Weight and Balance Data 2. Power Calculations
Comments (Good and Bad): <u>DID NOT USE</u>
NCS Navigation Objective: To complete an AQS-14 mission using the Nav/Comm system as the primary means of navigation and mission system information. Use all available HSDS/CDU screens and pages as you see fit to assist in mission accomplishment.
Mission Narrative (General comments including minefield and tracks flown): Departed point X-RAY, did Mission Waypoints out to field. Used the Stream point set up by MPS. Completed
3/2 tracks. Recovered. Used MISSION waypoints to Oceana >
NGU.
1. In which ways did NCS aid in your ability to successfully conduct an AMCM mission? Describe in detail. NCS greatly enhanched mission situational awareness. Hoving copilet
up tow screen allows him to help pilot make better turns and
use piccise stream points. INCREASES OF TON time. Able to
plan & fly a very structured mission each time.
2. In which ways did NCS hinder your ability to successfully conduct an AMCM mission? Describe in detail. NCS does not really hinder ability to Successfully

Do you foresee gration?	any	AMCM	tactical	changes	as	a	result	of	NCS
	<u> </u>							<u>.</u>	

Date: ZLAPRDTM Sortie Name: ZLAPRONYOZ
AHAC CO-Pilot
Total Flight Time: 2.5 Tow Hrs 1.8 op Tow 1.0
MPS Performance Calculations Objective: To evaluate the effectiveness/accuracy of the MPS provided performance calculations.
1. Weight and Balance Data 2. Power Calculations ———
Comments (Good and Bad): Same Comments As Before
NCS Navigation Objective: To complete an AQS-14 mission using the Nav/Comm system as the primary means of navigation and mission system information. Use all available HSDS/CDU screens and pages as you see fit to assist in mission accomplishment.
Mission Narrative (General comments including minefield and tracks flown): Went to Make Creed ARE FLEW TRACKS R575, R151, R697, R21
L575 + 1/2 of R333. Q14 SMORRED SOMETHENG AND WE HAD TO RELOVET
Manually Sequences THROUGH REMOTORISE PSEQ.
1. In which ways did NCS aid in your ability to successfully conduct an AMCM mission? Describe in detail. MAR DESPLAY, DSF + XTK ARE EXCELLENT FOR SETURITARY AWAREINE
2. In which ways did NCS hinder your ability to successfully conduct an AMCM mission? Describe in detail.

	Data sheet E-3 Page 2 of 3
Did the NCS provide adequate steering cues avigate precisely in the minefield? ES NO If no, why?	
. Which NCS functions, if any, were particular onsuming, or required too much focus in the company of the subsect of the subse	COCKDIL:
. Were there any failures or difficul ncountered during the flight? YES NO Descr	ties with the NCS
. Did you see any abnormal CWA's?	
. Was critical mission information readily Time + Fom is NOT Readily AVARLABLE	available? . Espectallo Dozrat ML Sug
3. Did you deviate from your MPS created flight	ht plan? If yes, why

9. Do you foresee any AMCM tactical changes	as a result of NCS
integration? Sector Nomen CLATURE & TYPE OF	SECTOR LNAMES



Data Sheet E-3 Page 1 of 3

Date: 1/APK Q1401
AHAC Co-Pilot
Total Flight Time: 1.8 Tow Hrs Op Tow
MPS Performance Calculations Objective: To evaluate the effectiveness/accuracy of the MPS provided performance calculations.
1. Weight and Balance Data 2. Power Calculations
Comments (Good and Bad): No PROBLEM;
NCS Navigation Objective: To complete an AQS-14 mission using the Nav/Comm system as the primary means of navigation and mission system information. Use all available HSDS/CDU screens and pages as you see fit to assist in mission accomplishment.
Mission Narrative (General comments including minefield and tracks flown): Departed X-RAY FLEW NCS to Stream point.
Utilized MPS Stream point. Completed several
tracks. Mission about due to 0-14 Props.
RTB NO fl+ plan rouk.
1. In which ways did NCS aid in your ability to successfully conduct an AMCM mission? Describe in detail. Vastly increased Situational awareness. Following
For some reason, we lust the PSED (it segrenched to last track)
Tow screen still allowed me to navigate
2. In which ways did NCS hinder your ability to successfully conduct an AMCM mission? Describe in detail. Problem With Town body position

8. Did you deviate from your MPS created flight plan? If yes, why and how?

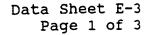
We flew our route axactly at flight plans

Do you egration	any	AMCM	tactical	changes	as	a	result	of	NCS

· 300 30

Date: 7 APROS DTM Sortie Name: 18 APROJUCT AHAC CO-Pilot
Total Flight Time: 2.8 Tow Hrs 2.1 Op Tow 1.5
MPS Performance Calculations Objective: To evaluate the effectiveness/accuracy of the MPS provided performance calculations.
1. Weight and Balance Data 2. Power Calculations
Comments (Good and Bad): No Pasciens. Within 190 of Calculated Value
NCS Navigation Objective: To complete an AQS-14 mission using the Nav/Comm system as the primary means of navigation and mission system information. Use all available HSDS/CDU screens and pages as you see fit to assist in mission accomplishment.
Mission Narrative (General comments including minefield and tracks flown): Flew our NCS flight plan to W50. Street
Mission Narrative (General comments including mineries and classes flown): Flew our NCS flight plan to W50. Streamed using MPS stream point. Completed 9 tracks. R-TB using NCS flight plan.
1. In which ways did NCS aid in your ability to successfully conduct an AMCM mission? Describe in detail.
Increased situational awareness allowed crow to pick an aptimum stream point. Occatation in turns
make for faster turns.
2. In which ways did NCS hinder your ability to successfully conduct an AMCM mission? Describe in detail.

9. Do you foresee integration?	any	AMCM	tactical	changes	as	a	result	of	NCS
			-						





Date: 18APR 95 DTM Sortie Name: 18 APR Q1407
AHAC Co-Pilot
Total Flight Time: 2.8 Tow Hrs Op Tow
MPS Performance Calculations Objective: To evaluate the effectiveness/accuracy of the MPS provided performance calculations.
1. Weight and Balance Data 2. Power Calculations
Comments (Good and Bad): 1/t + Balance cales were petty
much sight on.
NCS Navigation Objective: To complete an AQS-14 mission using the Nav/Comm system as the primary means of navigation and mission system information. Use all available HSDS/CDU screens and pages as you see fit to assist in mission accomplishment.
Mission Narrative (General comments including minefield and tracks
flown): Completed several tracks and concerned
flown): Completed several tracks and concerned
flown): Completed several tracks and collected 9 mark points. Mission was Very Successful for NCS, pilots and arriver.
flown): Completed several tracks and conceived
9 mark points. Mission was Very Successful for NCS, pilots and access. 1. In which ways did NCS aid in your ability to successfully conduct an AMCM mission? Describe in detail. Manelell O viewlation and Situational Awareness.
1. In which ways did NCS aid in your ability to successfully conduct an AMCM mission? Describe in detail. Murefield orientation and Strutumal Awareners Murefield over current techniques immensely.
1. In which ways did NCS aid in your ability to successfully conduct an AMCM mission? Describe in detail. Murefield orientation and Strutumal Awareners Murefield over current techniques immensely.
1. In which ways did NCS aid in your ability to successfully conduct an AMCM mission? Describe in detail. Murefield orientation and Strutumal Awareners Murefield over current techniques immensely.
9 mark points. Mission was Very Successful for NCS, pilots and access. 1. In which ways did NCS aid in your ability to successfully conduct an AMCM mission? Describe in detail. Manelell O viewlation and Situational Awareness.

9. Do you foresee any AMCM tactical changes as a result of N	CS
integration? lot less time lost in turns and	
stream to the field time. Who flight records we	<u>U</u>
show all A/L info at any time.	



Date: 1747e95 DTM Sortie Name:
AHAC Co-Pilot
Total Flight Time: 18 Tow Hrs 1.0 Op Tow .3
MPS Performance Calculations Objective: To evaluate the effectiveness/accuracy of the MPS provided performance calculations.
1. Weight and Balance Data 2. Power Calculations
Comments (Good and Bad): SAME COMMENTS AS BEFORE ABOUT STABLE
ENGENE, AND ABOUT THE WAY PERFORMENCE CALCULATIONS ARE DON'T
NCS Navigation Objective: To complete an AQS-14 mission using the Nav/Comm system as the primary means of navigation and mission system information. Use all available HSDS/CDU screens and pages as you see fit to assist in mission accomplishment.
Mission Narrative (General comments including minefield and tracks flown): NO PROBLEMS TO GET TO STREAM PT. MANUALLY SEG ONCE POST
STREAM PT TO GET STEETENG TO 1ST TEACH, Switches BACK TO AUTO Seq.
DURING THE STEED NOS SCOUENCED THROUGH ALL THE TRACKS EVERT THE LAST
ONE.
1. In which ways did NCS aid in your ability to successfully conduct an AMCM mission? Describe in detail. THE TOW PAGE GIVE GREAT SITUATIONAL AWARENESS, WETH THE
ADDITION OF ATT + DIST FROM GIVES GREAT STEATING CUES
2. In which ways did NCS hinder your ability to successfully conduct an AMCM mission? Describe in detail.
When IT EYELDS ALL THE WAY THOUGH TO THE LAST TRACK I HAD

Data Sheet E-3 Page 3 of 3

			resee	any	AMCM	tactica	l change	s as	a res	sult	of	NCS
inte	egra	tion	JUST	Lesa	TEME	SPENT (LANN FRO	+ PMA	THE	FLEG	+13	<i>ل</i> دين
1000		 AVATL		-								

Post-it" Fax Note 7671 Date pages 5

Co./Dept Co. New P.G. School

Phone # Phone #

Measurements

- A) Distance to middle of the TSI lighted LCD scale + 13/16 5 1/2"
- B) Distance to the middle of the TSI numerical readout 5 1/4
- C) Distance to middle of the TSI's skew bar 15/16 5
- D) Distance to GSDA Ground Speed output 3 3/4
- E) Distance to GSDA arrow center 334
- F) Distance to middle of the HSI 5 1/8"
- G) Distance to HSI's Course indication 3 1/4

H) Distance to HSI's miles indication 3 7/4

FROM

I) Distance to the middle of the mode select panel 8 1/4

PART 2

This part of the measurements I'm trying to get an idea of the approximate viewing areas of the ADI, TSI and HSI, see FIG 2.

- 1) ADI dimensions A1 4 1/8 & A2 3 1/12
- 2) TSI dimensions B1 2 1/16 & B2 2 1/16
- 3) HSI dimensions C1 4 7/8 & C2 3 7/16

PART 3.

For this next part you probably will need some assistance. I need you or someone you want to measure who is within normal pilot height/wt requirements to sit in the pilot seat in a normal flying position. The measurements need to be:

- 1) From the cockpit floor to the pilot's eye height. 41 34
- 2) Approximate distance from the pilot's eye to the center of the ADI 30"
- 3) An estimate of the pilot's normal seat position

(i.e. full back & full down, 3 clicks back & 1 click up, etc)

fill back & fell den

- 4) With the pilot sitting on a hard surface (i.e. desk) and measure from the desk to the pilots eye height 32"
- 5) Approx distance from the pilot's eye to center of VO-30 screen 38/4
- 6) Approx distance from the co-pilot's eye to center of VO-30 screen 38
- 7) Pilot's estimated HT 7/ & WT 162

As a last request if you have any pictures of the VO-30 and a natops like write up on it's functions this would be greatly appreciated.

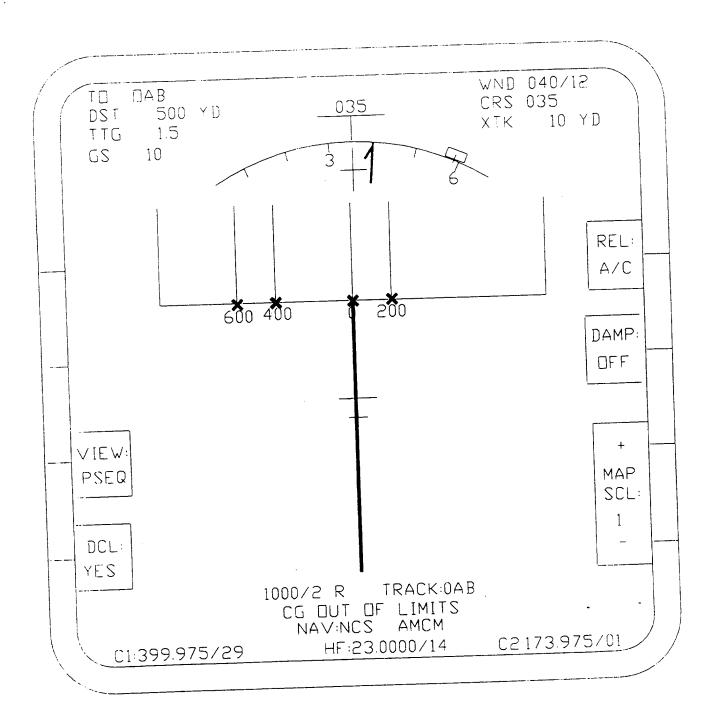
A huge thank to your & who ever you drafted into helping me out with my thesis.

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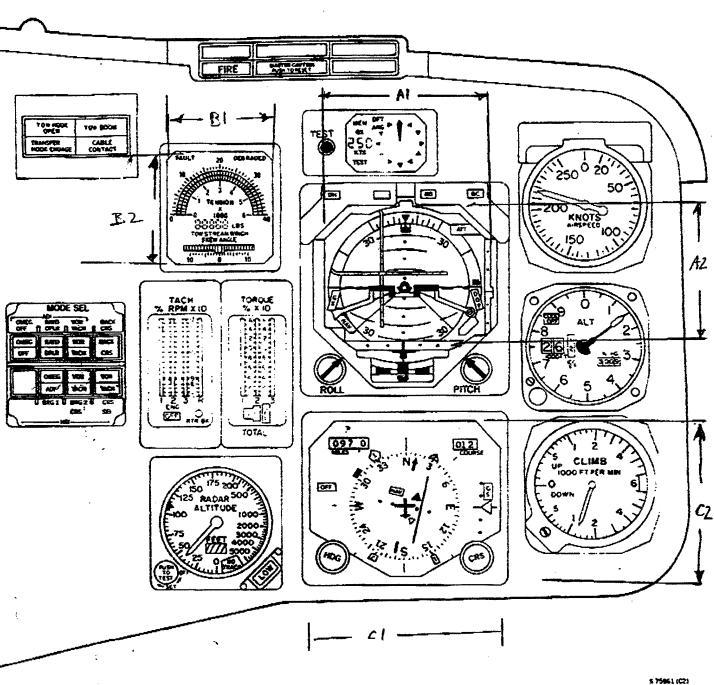
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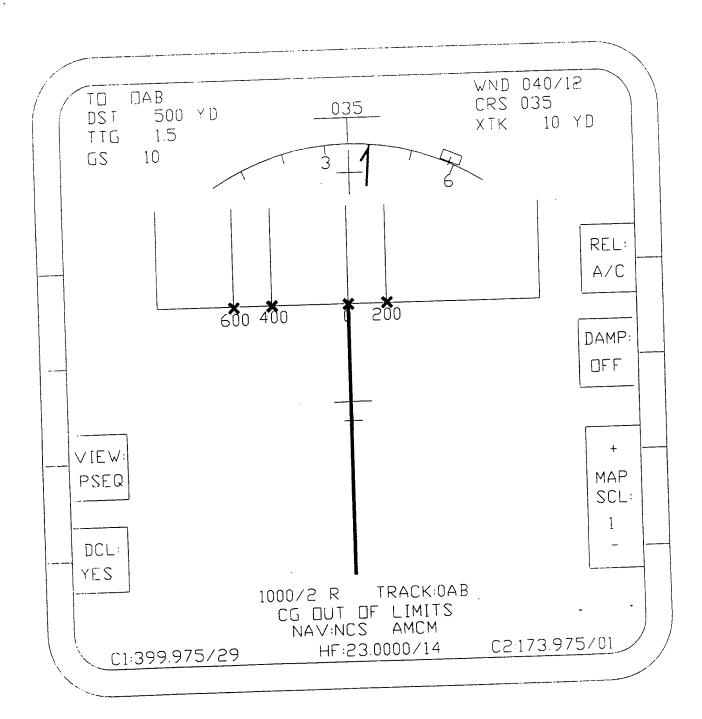




INSTRUMENT PANEL

FO-2 (Reverse Blank) CHANGE 3

F16 2





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